ORIGINAL PAPER



# A splitting of collinear libration points in circular restricted three-body problem by an artificial electrostatic field

4 Vladimir S. Aslanov

5 Received: 24 December 2020 / Accepted: 16 January 2021
6 © The Author(s), under exclusive licence to Springer Nature B.V. part of Springer Nature 2021

7 Abstract This paper focuses on the study of a new 8 type of a planar, circular restricted three-body problem 9 with an attractive artificial electrostatic field (E-field) 10 at collinear libration points. For instance, this attrac-11 tive field can be generated by an orbiting spacecraft 12 located at the Mars-Phobos L1 libration point and an 13 electrostatic capsule launched from Phobos. The 14 feasibility of the proposed retrieval system is dis-15 cussed from the aspect of local space weather Debye length. The attractive E-field splits the collinear 16 17 libration point into two new collinear points, and the 18 greater the E-field potential and the Debye length the 19 greater the distance between the new libration points 20 and the "old" original libration point. The new 21 equilibrium positions caused by the action of the 22 E-field have been found, and an instability of these 23 new libration points has been proven. A new Jacobi 24 integral in analytical form is obtained and equations of 25 motion are derived for the restricted problem of three 26 bodies taking into account the E-field. A numerical 27 simulation shows the impact of the E-field potential on 28 capsule capture in the small vicinity of the Mars-29 Phobos L1 libration point. This work expands the 30 classic three-body problem filling with new content. 31 The obtained results can be applied, for example, to

A1 V. S. Aslanov (🖂)

study an opportunity of delivering the Phobos samples32using Coulomb interaction of bodies in space.33

KeywordsLibration points · Electrostatic field ·34Stability and instability · Exact solutions · Jacobi35integral36

## 1 Introduction

The planar circular restricted three-body problem is a 38 classic celestial mechanics issue. Its study made a 39 great contribution to the theory of space dynamics and 40 celestial mechanics. The restricted three-body prob-41 lem is the basis for solving many applied tasks in 42 astronautics, in particular, for calculating interplane-43 tary flights and launching spacecraft and satellites. 44 This problem and its various aspects received great 45 attention from the scientific community. A detailed 46 analysis of major studies on this topic can be found in 47 the Szebehely's textbook [1]. The book discusses the 48 regularization of the motion equations, manifold of the 49 50 states of motion, equilibrium positions, motion near these positions, application of Hamiltonian dynamics 51 methods to the restricted problem, its periodic orbits 52 and quantitative aspects. A detailed numerical analysis 53 of three-dimensional periodic halo orbits near colli-54 near libration points in the restricted three-body 55 problem was performed by Howell [2]. Analytical 56

🖄 Springer

37

1



A2 Samara National Research University, 34, Moscovskoe

A3 shosse, Samara, Russia 443086

A4 e-mail: aslanov\_vs@mail.ru

A5 URL: http://aslanov.ssau.ru/

57

58 points  $L_1, L_2$  were obtained by Luo et al. in [3] using 59 normalization method. These results were compared 60 with the solutions obtained by Lindstedt-Poincare 61 method. Numerical analysis of the planar circular 62 restricted three-body problem phase space and orbits 63 classification into three groups (bounded, escape and collisional) were performed by Zotos [4]. Halo, 64 65 Lyapunov and vertical orbits for elliptic restricted 66 three-body problem were studied based on resonant 67 motions in the circular problem by Ferrari and Lavagna [5]. Woo and Misra [6] investigated the 68 69 spacecraft motion in the vicinity of a binary asteroid 70 system as the circular restricted case. The asteroids 71 were considered as rigid bodies. Addition equilibrium 72 points were found numerically for some special cases. 73 Biggs and Negri [7] considered solar sail spacecraft 74 controlled motion within the circular restricted three-75 body problem. The spacecraft moves in the gravita-76 tional field of the Earth and the Moon, taking into 77 account the perturbation introduced by the solar 78 pressure on the sail. Alessi and Sánchez [8] presented 79 a semi-analytical approach, which is based on a 80 perturbation procedure, for study the three-dimen-81 sional motion of a negligible mass body in the circular 82 restricted three-body problem. At this point, it should 83 be emphasized that the above references are not an 84 exhaustive analysis of the literature on the considered 85 topic, which is very broad and includes a huge number of works, but they give some insight into this problem. 86 New engineering ideas require solutions of new 87 88 fundamental problems. For example, in recent years, 89 Martian sample return missions were actively dis-90 cussed by scientists [9]. The missions assume that a 91 sample capsule is launched from Mars surface using a 92 Mars Ascent Vehicle after samples are collected by a 93 rover [10]. As a solution to a deep-space docking 94 challenge, the ability to use electrostatic force to 95 capture the container was discussed in [9]. The idea of 96 an electrostatic capture of the capsule can be used also 97 in the case when an orbital spacecraft (orbiter) is 98 located, in contrast to [9], at one of the collinear 99 libration points.

solutions for Lissajous and halo orbits near collinear

100 If we talk about Mars, then it can be point  $L_3$ , and if 101 about Mars's moon, Phobos, then points  $L_1$  and  $L_2$  are 102 more suitable. The Phobos exploration is of indepen-103 dent importance. Phobos is a small, irregularly shaped 104 moon ( $\sim 26 \times 22.8 \times 18.1$  km) that orbits Mars 105 every 7 h and 39 min. The orbit is synchronous to its

🖉 Springer



Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
Article No. : 6226	□ LE	□ TYPESET
MS Code : NODY-D-20-02999R1	🖌 СР	🖌 DISK

rotation so that its long axis is always directed toward 106 Mars. The Mars-Phobos  $L_1$  libration point is unusually 107 close to Phobos' surface ( $\sim 3.4$  km), and it make the 108 capsule delivery mission to a Phobos Sample Return 109 orbiter technically feasible if the orbiter hovered in the 110 Mars-Phobos  $L_1$  point. Several theoretical questions 111 have to be answered in order to investigate a 112 possibility of using the attractive artificial E-field at 113 the collinear libration points  $L_1$ ,  $L_2$  and  $L_3$ , for 114 example, the Phobos sample capsule delivery mission. 115

The answers to these questions are covered in this 116 article. This study focuses on equations and analytical 117 formulas describing the new planar circular restricted 118 three-body problem with the additional E-field. The 119 goal is to understand the body's behavior in the E-field 120 and the two gravitational fields caused by the main 121 bodies near the collinear libration points. Note that 122 when considering the E-field, the Debye length  $\lambda_D$ 123 must be taken into account. The Debye length is an 124 important parameter because the E-field rapidly 125 decreases beyond this length by the Debye shielding 126 effect. The motion equations are written in a rotating 127 Cartesian coordinate system, which is converted to a 128 canonical dimensionless form. We show the splitting 129 of the "old" collinear liberation points  $L_1, L_2, L_3$ , and 130 find new paired equilibrium positions  $(L_{1+}, L_{1-}; L_{2+}, L_{2+})$ 131  $L_{2-}$ ;  $L_{3+}, L_{3-}$ ) in the vicinity of the "old" points, and 132 also prove the instability of these new equilibrium 133 positions. Next, a new Jacobi integral is obtained in 134 analytical form for the circular restricted three-body 135 problem taking into account the E-field. And finally, 136 using numerical modeling, we show the effect of the 137 E-field potential on the capture of the electrostatic 138 capsule (E-capsule) in the small vicinity of the Mars-139 Phobos  $L_1$  libration point. 140

## 2 Equations of motion and Jacobi integral

141

142 In this section, we derive the planar motion equations 143 of a body in two gravitational and one attractive Efield. The center of E-field is located at a single 144 collinear libration point. These points are called the 145 "old" collinear libration points because, as can be 146 assumed (this will be proved below), due to the action 147 of the E-field, the "old" unstable point splits into two 148 new unstable equilibrium points. The action of the E-149 fields is limited to the Debye length within tens of 150 meters. In addition, the motion of a body is studied 151

152 only in a small vicinity of "old" points comparable to 153 Debye length, beyond which the problem degenerates into the classic three-body problem. Therefore, we talk 154 155 about two gravity fields and one attractive E-field, and the motion equations are written for the vicinity of 156 157 each point:  $L_1$ ,  $L_2$ ,  $L_3$ . The equations of motion are 158 derived using classical terminology of the three-body 159 problem presented in book by Schaub and Junkins 160 [11]. This will facilitate understanding of the pre-161 sented material. Below this section is organized as 162 follows. Firstly, the conventional assumptions for the circular restricted three-body problem are described. 163 Next, the expression for Coulomb force acting on the 164 charged body in the E-field is given. Then, the 165 equations of the body motion are derived in canonical 166 form. Finally, the non-dimensional potential function 167 168 and the Jacobi integral are found.

169 The considered mechanical system consists of three bodies  $M_1$ ,  $M_2$  and M(Fig. 1). It is assumed that the 170 171 mass of the body M is many times less than the mass of 172 the bodies  $M_1$  and  $M_2$ . Therefore, the body M has 173 negligible effect on the other bodies. It is also 174 supposed that the bodies  $M_1$  and  $M_2$  move in circular 175 orbits around their mutual center of mass. Before 176 proceeding with the construction of the relative 177 motion equations, the electrostatic force is introduced. 178 The orbiter is assumed to be located at one of the 179 collinear libration points of  $M_1$ -  $M_2$  bodies system and 180 to have an electrostatic charge. The mass of the orbiter is not taken into account. The charged electrostatic 181 182 capsule (E-capsule) is affected by an electrostatic 183 force when moving in the vicinity of the orbiter. The 184 capsule performance is dependent on this force. It is assumed that the capsule and the orbiter are perfectly 185



Fig.1 The Local-Vertical-Local-Horizontal frame Oxy

conducting spheres. The electrostatic force between 186 the capsule and orbiter is defined by the equation 187

$$F = k_C \frac{q_O q}{R^2} \tag{1}$$

where  $k_C = 8.99 \cdot 10^9 N \text{m}^2 / C^2$  is the Coulomb con-189 stant,  $q_0$  is the charge on the orbiter, q is the charge on 190 the capsule, R is the distance between the capsule and 191 orbiter. Note that this force may be either attractive or 192 repulsive depending on the polarity of the potentials of 193 the E-capsule and orbiter. Consider only the attracting 194 configuration  $(q_0 q < 0)$  in this paper. 195

Let us first introduce into consideration three 196 vectors in the Local-Vertical-Local-Horizontal frame 197 Oxy: the position vector of the body M (Fig. 1) 198

$$\mathbf{r} = (r_x, r_y) \tag{2}$$

the vector  $\mathbf{a} = (a, 0)$  determines the position of the 200 collinear libration points  $L_1$ ,  $L_2$  and  $L_3$  in the frame 201 Oxy 202

$$a = (a_1, a_2, a_3), \quad a_i = \left|\overline{OL_i}\right|, \ i = 1, 2, 3$$
 (3)

and, the position vector of the body M with respect 204 to the collinear libration point (Fig. 1) 205

$$\mathbf{R} = \mathbf{r} - \mathbf{a} = (X = r_x - a, Y = r_y)$$
(4)

Then, the vector of the Coulomb force (1) is written 207 208 as

$$\mathbf{F} = kF \frac{\mathbf{R}}{R} = k \frac{k_C q_O q}{R^3} \mathbf{R}, \quad (q_O q < 0)$$
(5)

Note that the Coulomb force affects the body M, if 210 this body is within the Debye length of a collinear 211 libration point, i.e., within 10 s meters 212

$$k = \begin{cases} 1, \ \sqrt{(r_x - a)^2 + r_y^2} \le \lambda_D \\ 0, \ \sqrt{(r_x - a)^2 + r_y^2} > \lambda_D \end{cases}$$
(6)

Now, taking into account the Coulomb force (5), 214 the equations of the E-capsule planar motion can be 215 written the frame *Oxy*[11] 216

$$\ddot{r}_x = \frac{\partial W}{\partial r_x} + \omega^2 r_x + 2n\dot{r_y} + k\frac{k_C q_O q}{m} \frac{(r_x - a)}{R^3}$$
(7)

$$\ddot{r}_y = \frac{\partial W}{\partial r_y} + \omega^2 r_y - 2n\dot{r}_x + k \frac{k_C q_O q}{m} \frac{r_y}{R^3}$$
(8)

Springer

	Journal : Medium 11071
	Article No. : 6226
$\sim$	MS Code : NODY-D-20-02

~	Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
	Article No. : 6226	□ LE	□ TYPESET
	MS Code : NODY-D-20-02999R1	🗹 СР	🗹 disk

$$W(r_x, r_y) = G\left(\frac{m_1}{\sqrt{(r_x + d\mu)^2 + r_y^2}} + \frac{m_2}{\sqrt{(r_x - d(1 - \mu))^2 + r_y^2}}\right)$$
(9)

where  $\mu = \frac{m_2}{m_1+m_2}$ ;  $\omega$  is a constant angular velocity magnitude of the  $M_1-M_2$  system; *d* is the distance between  $M_1$  and  $M_2$ ; *m*,  $m_1$  and  $m_2$  are masses of the bodies *M*,  $M_1$  and  $M_2$ , respectively.

The motion equations (7) and (8) can be written in a convenient non-dimensional form. To do so, we introduce the non-dimensional time variable t as

$$\tau = \omega t \tag{10}$$

230 Time derivatives with respect to this new time231 variable are denoted as

$$(.)' = \frac{d}{d\tau}(.) \tag{11}$$

233 Any scalar distances are non-dimensionalized by 234 dividing them with the constant relative between  $M_1$ 235 and  $M_2$  as

$$x = \frac{r_x}{d}, \ y = \frac{r_y}{d} \tag{12}$$

Using the variables substitutions (10) and (12), we
now are able to rewrite Eqs. (7) and (8) into the
following non-dimensional form:

$$x'' - 2y' = \frac{\partial U}{\partial x} \tag{13}$$

$$y'' + 2x' = \frac{\partial U}{\partial y} \tag{14}$$

243 where  $\alpha = \frac{a}{d} < 1$ . The corresponding non-dimensional 244 potential function U(x, y) is given by the expression

$$U(x,y) = \frac{1}{2}(x^{2} + y^{2}) + \frac{1 - \mu}{\sqrt{(\mu + x)^{2} + y^{2}}} + \frac{\mu}{\sqrt{(-1 + \mu + x)^{2} + y^{2}}} - k\frac{\Phi}{\sqrt{(x - \alpha)^{2} + y^{2}}}$$
(15)

246 where

$$\Phi = \frac{k_C q_O q}{m d^3 \omega^2} < 0 \tag{16}$$

Due to the definition of the mass ratio  $\mu$  the nondimensional coordinates of  $M_1$  and  $M_2$  are written as. 249

$$x_1 = -\mu, \quad x_2 = 1 - \mu$$
 (17)

This expression is simplified by entering a nondimensional relative distance  $\rho_i$  is defined as 252

$$\rho_i = \sqrt{(x - x_i)^2 + y^2}, \quad \rho = \sqrt{(x - \alpha)^2 + y^2} \quad (18)$$

Then, the corresponding non-dimensional potential254function (15) is given by the expression255

$$U(x,y) = \frac{1}{2}(x^2 + y^2) + \frac{1-\mu}{\rho_1} + \frac{\mu}{\rho_2} - k\frac{\Phi}{R}$$
(19)

Following similar steps as were done with the 257 classic restricted three-body problem [11], with the Efield the non-dimensional Jacobi integral takes on the form 260

$$T(x,y) = (x^{2} + y^{2}) + \frac{2(1-\mu)}{\sqrt{(\mu+x)^{2} + y^{2}}} + \frac{2\mu}{\sqrt{(-1+\mu+x)^{2} + y^{2}}} - 2k\frac{\Phi}{\sqrt{(x-\alpha)^{2} + y^{2}}} - (x'^{2} + y'^{2})$$
(20)

For the dimensional equations of motion (13) and (14) the non-dimensional Jacobi integral is written as 263

$$J(x,y) = (x^{2} + y^{2}) + \frac{2(1-\mu)}{\rho_{1}} + \frac{2\mu}{\rho_{2}} - 2k\frac{\Phi}{\rho} - (x'^{2} + y'^{2})$$
(21)

265

266

#### **3** New Lagrange collinear libration points

Setting the relative velocities and accelerations in 267 Eqs. (13) and (14) equal to zero, we find conditions 268 that are satisfied by the stationary points of the circular 269 restricted three-body problem with the E-field. Obvi-270 ously, the new stationary points can be located only 271 within the Debye length of the "old" collinear 272 libration points  $L_i$  (i = 1, 2, 3), i.e., within tens of 273 meters. Outside this boundary, this task degenerates 274 into the classic circular restricted three-body problem 275 when, k = 0 according to Eqs. (6), (13) and (14). 276

Springer



•	Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
	Article No. : 6226	🗆 LE	□ TYPESET
	MS Code : NODY-D-20-02999R1	🗹 СР	🖌 disk

222

223 224

#### 277 Therefore, we will look for new libration points only 278 inside the Debye sphere is defined as

$$\sqrt{\left(r_x - a\right)^2 + r_y^2} \le \lambda_D \tag{22}$$

280 or, taking into account (12)

$$\sqrt{\left(x-\alpha\right)^2+y^2} \le \frac{\lambda_D}{d} = l_D \tag{23}$$

To solve for the scalar coordinate x for the collinear libration points, Eq. (13) is set equal to zero for y' = y = 0, simplify to following [11]

$$0 = x - \frac{\mu(-1+\mu+x)}{(-1+\mu+x)^3} - \frac{(1-\mu)(\mu+x)}{(\mu+x)^3} + k \frac{\Phi(x-\alpha)}{(x-\alpha)^3}$$
(24)

286

282

283

284

Author Proo

Let's rewrite Eq. (24) with regard to (17) as

$$0 = x - \frac{\mu(x - x_2)}{(x - x_2)^3} - \frac{(1 - \mu)(x - x_1)}{(x - x_1)^3} + k \frac{\Phi(x - \alpha)}{(x - \alpha)^3}$$
(25)

288 Using the fact that the new libration points gener-289 ated by the E-field must be located within the Debye 290 sphere (23), we begin with the vicinity of the point  $L_1$ 291 and find from Eq. (25) implicit conditions for the  $L_{1-}$ and  $L_{1+}$  position coordinates in terms of the mass ratio 292 293 μ:

$$L_{1-}: 0 = x - \frac{1-\mu}{(\mu+x)^2} + \frac{\mu}{(x-1+\mu)^2} - \frac{\Phi}{(x-\alpha)^2}$$
(26)

295

$$L_{1+}: 0 = x - \frac{1-\mu}{(\mu+x)^2} + \frac{\mu}{(x-1+\mu)^2} + \frac{\Phi}{(x-\alpha)^2}$$
(27)

297 The last terms differentiate these equations from 298 Eq. (10.93) in [11] and show the splitting of the point 299  $L_1$  into two new unstable collinear points  $L_{1-}$  and  $L_{1+}$ . 300 Now consistently, in a similar way, we write down the 301 implicit conditions for the collinear libration points in 302 the vicinity of the "old" libration points  $L_2$  and  $L_3$ :

$$L_{2-}: 0 = x - \frac{1-\mu}{(\mu+x)^2} - \frac{\mu}{(x-1+\mu)^2} - \frac{\Phi}{(x-\alpha)^2}$$
(28)

$$L_{2+}: 0 = x - \frac{1 - \mu}{(\mu + x)^2} - \frac{\mu}{(x - 1 + \mu)^2} + \frac{\Phi}{(x - \alpha)^2}$$
(29)

and

$$L_{3-}: 0 = x + \frac{1-\mu}{(\mu+x)^2} + \frac{\mu}{(x-1+\mu)^2} + \frac{\Phi}{(x-\alpha)^2}$$
(30)

306

312

$$L_{3+}: 0 = x + \frac{1-\mu}{(\mu+x)^2} + \frac{\mu}{(x-1+\mu)^2} - \frac{\Phi}{(x-\alpha)^2}$$
(31)

The order of the location of the new libration points 310 caused by the E-field is given in Fig. 2. 311

#### 4 Libration points stability

To study the stability of the new equilibrium positions 313  $(L_{1+}, L_{1-}, L_{2+}, L_{2-}, L_{3+}, L_{3-})$  caused by the E-field, 314 we use a standard linearization procedure [12]. Firstly, 315 the equations of relative motion are linearized in the 316 small vicinity of the new equilibrium position. Then, 317 the eigenvalues of the linearized plant matrix are 318 determined. The conclusion about the equilibrium 319 position stability is made on the basis of the real part of 320 these eigenvalues. Let denote the position of the 321 Lagrange point as  $(x_0, y_0)$ . The body *M* is located at the 322 point with coordinates  $(x_0 + \xi, y_0 + \eta)$ . Calculating 323 the body velocity components  $(\xi', \eta')$ , substituting 324 these quantities into Eqs. (13), (14) and expanding 325 327 326 result in a Taylor series gives

$$\xi'' - 2\eta' = \xi \frac{\partial^2 U}{\partial x^2} \bigg|_0 + \eta \frac{\partial^2 U}{\partial x \partial y} \bigg|_0 + \cdots$$
(32)

$$\eta'' + 2\xi' = \xi \frac{\partial^2 U}{\partial x \partial y} \bigg|_0 + \eta \frac{\partial^2 U}{\partial y^2} \bigg|_0 + \cdots$$
(33)

where the suffix zero means that after the partial If the 331 displacements  $\xi$  and  $\eta$  are small, we may neglect terms 332 involving squares, products and higher-degree terms 333 335 334 in  $\xi$  and  $\eta$ , and so the equations become

$$\xi'' - 2\eta' = \xi U_{xx} + \eta U_{xy} \tag{34}$$

$$\eta'' + 2\xi' = \xi U_{yx} + \eta U_{yy} \tag{35}$$

where

Journal
Article
MS Co

Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
Article No. : 6226	□ LE	□ TYPESET
MS Code : NODY-D-20-02999R1	🗹 СР	🖌 DISK



**Fig. 2** Splitting the collinear libration points, when  $\Phi = \frac{k_C q_O q}{md^3 \omega^2} = -9 \cdot 10^{-16}$ 

$$U_{xx} = \frac{\partial^2 U}{\partial x^2} \Big|_0, \quad U_{yy} = \frac{\partial^2 U}{\partial y^2} \Big|_0,$$
  

$$U_{xy} = U_{yx} = \frac{\partial^2 U}{\partial x \partial y} \Big|_0 = \frac{\partial^2 U}{\partial y \partial x} \Big|_0$$
(36)

and the U are constant since they are evaluated at the
Lagrange point. These are linear differential equations
with constant coefficients, the general solution of
which may be written as

$$\xi = \sum_{i=1}^{4} \alpha_i \exp(\lambda_i t), \quad \eta = \sum_{i=1}^{4} \beta_i \exp(\lambda_i t)$$
(37)

346 where  $\alpha_i$  are integration constants, the constants  $\beta_i$ 

Deringer

Author Proof



dependent upon $\alpha_i$ and the constants appearing in the	347
differential equations. The $\lambda_i$ are the roots of the	348
characteristic determinant of Eqs. (34) and (35) set	349
equal to zero and rewritten as	350

$$\begin{vmatrix} \lambda^2 - U_{xx} & -2\lambda - U_{xy} \\ 2\lambda - U_{xy} & \lambda^2 - U_{yy} \end{vmatrix} = 0$$
(38)

or

$$\lambda^{4} + (4 - U_{xx} - U_{yy})\lambda^{2} + U_{xx}U_{yy} - U_{xy}^{2} = 0$$
 (39)

The solution is stable when all  $\lambda_i$  obtained from 354 Eq. (39) are pure imaginary numbers. Since, along 355 with any root  $\lambda$ , the biquadratic characteristic equation 356 also has a root  $-\lambda$ , then the solution is unstable when 357

Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
Article No. : 6226	□ LE	□ TYPESET
MS Code : NODY-D-20-02999R1	🖌 СР	🗹 disk

(41)

any of the  $\lambda_i$  are real or complex numbers with a nonzero real part. Now, defining the following quantities *A*, *B*, *C*, *D* and *E* as

$$A = \frac{1 - \mu}{\rho_1^3} + \frac{\mu}{\rho_2^3} \tag{40}$$

$$B = 3(\frac{1-\mu}{\rho_1^5} + \frac{\mu}{\rho_2^5})$$

$$C = 3\left(\frac{1-\mu}{\rho_1^5}(x_0+\mu) + \frac{\mu}{\rho_2^5}(x_0-(1-\mu))\right)$$
(42)

$$D = k \frac{\Phi}{\left|\rho\right|^3} \tag{43}$$

368

364

366

Author Proo

$$E = k \frac{\Phi}{|\rho|^5} \tag{44}$$

We find that

and

$$U_{xx} = 1 - A + 3(1 - \mu) \frac{(x_0 + \mu)^2}{\rho_1^5} + 3\mu \frac{(x_0 - (1 - \mu))^2}{\rho_2^5} + D - 3E(\alpha - x_0)^2$$

372 
$$U_{yy} = 1 - A + By_0^2 + D - 3Ey_0^2$$
 (46)

$$U_{xy} = Cy_0 + 3E(\alpha - x_0)y_0 \tag{4}$$

376 In the straight line solution,  $y_0 = 0$ , so that

$$\rho_i^2 = (x_0 - x_i)^2, \quad \rho^2 = (x_0 - \alpha)^2$$
(48)

378 Hence

$$U_{xx} = 1 + 2A - 2D, \quad U_{yy} = 1 - A + D, \quad U_{xy} = 0$$
(49)

380 Applying the values from (49) in Eq. (39) we obtain

$$\lambda^4 + (2 - A')\lambda^2 + (1 + A' - 2A'^2) = 0$$
(50)

382 where

$$A' = A - D = \frac{1 - \mu}{\rho_1^3} + \frac{\mu}{\rho_2^3} - k\frac{\Phi}{\rho^3}$$
(51)

384 It can be shown that

$$1 + A' - 2A'^2 < 0 \tag{52}$$

for values of  $\mu$  up to its limit of  $\frac{1}{2}$ . Hence, the four roots386of equation (50) consist of two real roots, numerically387equal but opposite in sign, and two conjugate pure388imaginary roots. Hence the solution is unstable.389Obviously, condition (52) with quadratic dependence390on the left side is satisfied for any A' greater than unity391

$$A' = A - D > 1 \tag{53}$$

Let us demonstrate that all the new equilibrium 393 positions  $(L_{1+}, L_{1-}, L_{2+}, L_{2-}, L_{3+}, L_{3-})$  satisfy this 394 condition. Consider the first two of these points 395  $L_{1+}, L_{1-}$  located within the Debye sphere with respect 396 to the point  $L_1$ . Since these points are located between 397  $M_1$  and  $M_2$ , then  $\rho_1, \rho_2 < 1$  and therefore 398

$$A = \frac{1-\mu}{\rho_1^3} + \frac{\mu}{\rho_2^3} > 1 - \mu + \mu = 1$$
(54)

Now consider the parameter D, which in consequence of (43) and (1) within the Debye sphere is always negative: 402

$$D = k \frac{\Phi}{\left|\rho\right|^3} < 0 \tag{55}$$

Thus, the condition (53) is satisfied for the equilib-<br/>rium positions  $L_{1+}, L_{1-}$ , and hence the condition (52)404<br/>405is also satisfied, hence the equilibrium positions406<br/>406 $L_{1+}, L_{1-}$  are unstable.407

We now investigate the stability at the other collinear equilibrium positions  $L_{2+}$ ,  $L_{2-}$ ,  $L_{3+}$  and  $L_{3-}$ , found from the four equations (28)–(31). Below we prove that at these points A' > 1 as well. For this purpose equation (25) can be rewritten as 412

$$0 = x(1 - \frac{1 - \mu}{\rho_1^3} - \frac{\mu}{\rho_2^3}) + \frac{(1 - \mu)}{\rho_1^3}x_1 + \frac{\mu}{\rho_2^3}x_2 + k\frac{\Phi(x - \alpha)}{(x - \alpha)^3}$$
(56)

Considering that  $x_1 = -\mu$ ,  $x_2 = 1 - \mu$ , get

$$x(\frac{1-\mu}{\rho_1^3} + \frac{\mu}{\rho_2^3} - 1) = \mu(1-\mu)(\frac{1}{\rho_2^3} - \frac{1}{\rho_1^3}) + k\frac{\Phi(x-\alpha)}{(x-\alpha)^3}$$
(57)

comparing this expression with Eqs. (40), (43) and417(53) write it as418

1

Deringer



,	Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
	Article No. : 6226	□ LE	□ TYPESET
,	MS Code : NODY-D-20-02999R1	🗹 СР	🗹 disk

$$A' - 1 = \frac{1 - \mu}{\rho_1^3} + \frac{\mu}{\rho_2^3} - 1 - k \frac{\Phi}{|\rho|^3}$$
  
=  $\frac{\mu(1 - \mu)}{x} (\frac{1}{\rho_2^3} - \frac{1}{\rho_1^3}) - k \frac{\Phi}{|\rho|^3} + k \frac{\Phi(x - \alpha)}{x(x - \alpha)^3}$   
(58)

420 Note that at the points  $L_{2+}$  and  $L_{2-}$ , the conditions 421 x > 0 and  $\rho_2 < \rho_1$  are satisfied, and at the points  $L_{3+}$ 422 and  $L_{3-}$ , the conditions x < 0 and  $\rho_2 > \rho_1$ . Besides

$$-k\frac{\Phi}{|\rho|^3} + k\frac{\Phi(x-\alpha)}{x(x-\alpha)^3} > 0$$
(59)

since  $\Phi < 0$  and x always many orders of magnitude greater than  $\rho$ , which does not exceed the relative Debye length (23). Therefore, in these considered cases we get

$$A' > 1 \tag{60}$$

429 It follows that condition (52) is also satisfied. Thus, 430 we have proved that the all new collinear equilibrium 431 points  $(L_{1+}, L_{1-}, L_{2+}, L_{2-}, L_{3+}, L_{3-})$  are unstable.

## 432 **5** Numerical simulation

433 This section shows the effect of the electrostatic 434 charge level on the capture of the E-capsule in the 435 vicinity of the Mars-Fobos libration point  $L_1$ , in which 436 the orbiter together with the E-capsule generates the attracting E-field. Note that the Debye length near the 437 438  $L_1$  point is unknown; approximate Debye length 439 values are given for the Stickney crater, located on 440 Phobos' surface directly under the  $L_1$  point. Depending on Mars local time, this parameter ranges from 13 441 442 to 47 m [14]. In the absence of accurate data, all 443 calculations are performed for  $\lambda_D = 45m$ . Since we 444 study the motion of the capsule relative to the orbiter, 445 which is in the  $L_1$  point, it makes sense to pass from the 446 frame Oxy to the frame  $L_1XY$  (Fig. 3) by changing the 447 variable

$$x = X + a_1, y = Y$$
 (61)

449where  $a_1$  is abscissa of the Mars-Phobos L1 point450Then taking into account (61), the motion equations451(7) and (8) can be rewritten as

$$\ddot{X} = \frac{\partial W_E}{\partial X} + \omega^2 (X + a_1) + 2\omega \dot{Y}$$
(62)

$$\ddot{Y} = \frac{\partial W_E}{\partial Y} + \omega^2 Y - 2\omega \dot{X}$$
(63)

where the effective potential is written as

$$W_{E}(X,Y) = G\left(\frac{m_{1}}{\sqrt{(X+a_{1}+d\mu)^{2}+Y^{2}}} + \frac{m_{2}}{\sqrt{(X+a_{1}-d(1-\mu))^{2}+Y^{2}}}\right)$$
$$-k\frac{P}{m\sqrt{X^{2}+Y^{2}}}$$
(64)

where  $P = k_C q_O q < 0$  is the charge level.

All trajectories of the capsule m = 10 kg begin with458the same initial conditions in the vicinity of point  $L_1$ 459from the side of Phobos in the frame  $L_1XY$ 460

$$X_0 = 81.533 \, m, \quad Y_0 = 10.829 \, m, \dot{X}_0 = -0.043 \, m/s, \quad \dot{Y}_0 = -0.017 \, m/s$$
(65)

Figure 3 demonstrates the E-capsule trajectories 462 for different charge levels: 463

$$P = k_C q_O q = 0, -0.28, -0.32, -0.40 [N \cdot m^2]$$

In the first case the E-field is absent, therefore, the 465 capsule does not reach the vicinity of the point  $L_1$ , the 466 same thing we can see when the charge level is not 467 high enough ( $P = -0.28 \,\mathrm{N} \cdot \mathrm{m}^2$ ). If the charge level is 468  $P = -0.32 \,\mathrm{N} \cdot \mathrm{m}^2$ equal and more 469  $(P = -0.40 \,\mathrm{N} \cdot \mathrm{m}^2)$ , there are several turns of the 470 capsule around the point  $L_1$ , and the greater the charge 471 level, the more turns the E-capsule performs around 472 the point  $L_1$ . 473

## 6 Conclusion

This paper shows within the framework of the planar475circular restricted three-body problem that the artificial attractive E-field at one of the collinear libration476points causes the splitting of this point into two other478unstable collinear libration points located within the479Debye length. This fact was proved analytically, using480

Deringer



Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
Article No. : 6226	🗆 LE	□ TYPESET
MS Code : NODY-D-20-02999R1	🗹 СР	🖌 DISK

424

425

426

427

474



Fig. 3 The capsule trajectories and Contour plot of the effective potential  $W_E(X, Y)$  in the frame  $L_1XY$ 

481 the corresponding analytic equations, the new Jacobi 482 integral, and confirmed by the numerical simulation. 483 We have demonstrated the feasibility of the attracting 484 mission the E-capsule in a small vicinity of the Mars-Fobos libration point  $L_1$ . 485

Acknowledgements This study was supported by the Russian 487 Science Foundation (Project No. 19-19-00085).

#### 489 **Compliance with ethical standards**

Conflict of interest	The author declares that he has no con-	490
flict of interest.		491

#### References

493 1. Szebehely, V.: The Restricted Problem of Three Bodies. 494 Academic Press Inc., New York (1967)

Deringer



<b>S</b>

Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
Article No. : 6226	□ LE	□ TYPESET
MS Code : NODY-D-20-02999R1	🗹 СР	🖌 DISK

525

526

527

528

529

530

531

532

533

534

535

536

537

538

539

514

515

495

496

- Connor Howell, K.: Three-dimensional, periodic, 'halo' orbits. Celestial Mech. 32, 53–71 (1984). https://doi.org/10. 1007/BF01358403
- Luo, T., Pucacco, G., Xu, M.: Lissajous and halo orbits in the restricted three-body problem by normalization method. Nonlinear Dyn. 101, 2629–2644 (2020). https://doi.org/10. 1007/s11071-020-05875-1
- Zotos, E.E.: Classifying orbits in the restricted three-body problem. Nonlinear Dyn. 82, 1233–1250 (2015). https://doi. org/10.1007/s11071-015-2229-4
- Ferrari, F., Lavagna, M.: Periodic motion around libration points in the elliptic restricted three-body problem. Nonlinear Dyn. 93, 453–462 (2018). https://doi.org/10.1007/ s11071-018-4203-4
- Woo, P., Misra, A.K.: Equilibrium points in the full threebody problem. Acta Astronaut. 99, 158–165 (2014). https:// doi.org/10.1016/j.actaastro.2014.02.019
- Biggs, J.D., Negri, A.: Orbit-attitude control in a circular restricted three-body problem using distributed reflectivity devices. J. Guid. Control Dyn. 42(12), 2712–2721 (2019). https://doi.org/10.2514/1.G004493
- 8. Alessi, E.M., Sánchez, J.P.: Semi-analytical approach for
  distant encounters in the spatial circular restricted threebody problem. J. Guid. Control Dyn. 39(2), 351–359 (2016).
  https://doi.org/10.2514/1.G001237

- 9. Shibata, T., Bennett, T., Schaub, H.: Prospects of a hybrid magnetic/electrostatic sample container retriever. J. Spacecraft Rockets 57(3), 434–445 (2020). https://doi.org/10. 2514/1.A34509
  10. Dankanich, J., Klein, E.: Mars ascent vehicle development 524
- Dankanich, J., Klein, E.: Mars ascent vehicle development status, IEEE Aerospace Conference, Big Sky. USA (2012). https://doi.org/10.1190/AERO.2012.6187295
- Schaub, H., Junkins, J.: Analytical Mechanics of Space Systems, AIAA EducationSeries, 2nd edn. Reston, VA (2009)
- 12. Roy, A.E.: Orbital motion. IoP, Bristol and Philadelphia (2005)
- Hartzell, C.M., Farrell, W., Marshall, J.: Shaking as means to detach adhered regolith for manned Phobos exploration. Adv. Space Res. 62(8), 2213–2219 (2018). https://doi.org/ 10.1016/j.asr.2017.09.010
- Martynov, M.B., et al.: Planetary protection principles used for Phobos-Grunt mission. Sol. Syst. Res. 45–7, 593–596 (2011). https://doi.org/10.1134/S0038094611070185

Publisher's NoteSpringer Nature remains neutral with<br/>regard to jurisdictional claims in published maps and<br/>institutional affiliations.540<br/>541<br/>542



Journal : Medium 11071	Dispatch : 29-1-2021	Pages : 10
Article No. : 6226	□ LE	□ TYPESET
MS Code : NODY-D-20-02999R1	🖌 СР	🗹 disk